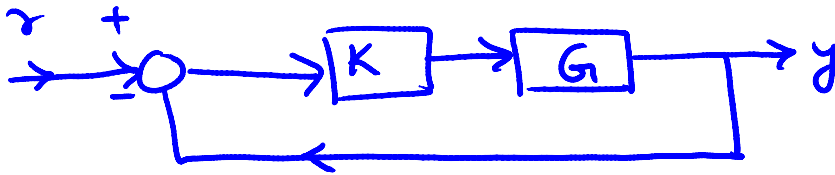


Bode-plot and stability

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Consider the unity negative feedback interconnection given below



We will first analyze the case when

K is a real and positive constant.

and thus $K(s) = k$ and thus $\frac{n_k}{d_k}$ with

$n_k = k$ and $d_k = 1$ forms a coprime representation of the controller K .

⊙ The characteristic polynomial is given by

$$n_G n_k + d_G d_k$$

$$= n_G k + d_G \quad \text{where } \frac{n_G}{d_G} \text{ is a}$$

coprime representation of G .

K goes to zero

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Thus, the stability of the feedback interconnection is governed by the roots of the polynomial given by

$$KN_G + d_G = 0$$

Clearly as $K \rightarrow 0$ the polynomial

$$KN_G + d_G \approx d_G(s)$$

and thus,

→ as $K \rightarrow 0$; the roots of the characteristic polynomial are given by the poles of the transfer function G (i.e. roots of d_G).

⊕ Now let's consider what happens when $K \rightarrow \infty$.

K goes to infinity

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The characteristic polynomial

$K N_G + d_G$ can be rewritten

as $N_G + \frac{1}{K} d_G$

and as $K \rightarrow \infty$

$N_G + \frac{d_G}{K} \approx N_G$ and

thus, the roots of the characteristic polynomial go toward the open-loop zeros of the plant G_1 (i.e. roots of N_G).

Also, in most cases

$$\deg(N_G) = m$$

$$\deg(d_G) = n$$

$$\text{with } n - m = q > 0$$

poles at infinity

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Thus, the characteristic polynomial roots are obtained by

$$d_G + K n_G = 0$$

$$\Leftrightarrow 1 + K \frac{n_G}{d_G} = 0$$

$$\Leftrightarrow \frac{n_G}{d_G} = -\frac{1}{K} \quad \text{-----} \quad (*)$$

clearly as $K \rightarrow \infty$; $-1/K \rightarrow 0$

Thus, for \bar{s} to be solution of (*)

$$\frac{n_G(\bar{s})}{d_G(\bar{s})} = 0$$

that happens at every root of $n_G(s)$.

However this condition can also hold as $s \rightarrow \infty$. Note that

as $s \rightarrow \infty$

poles at infinity

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and assuming

$$N_u(s) = \alpha (s^m + a_{m-1}s^{m-1} + \dots + a_0s^0)$$

$$D_u(s) = s^n + b_{n-1}s^{n-1} + \dots + b_0s^0$$

We have

$$\frac{N_u(s)}{D_u(s)} = \alpha \left[\frac{s^m + a_{m-1}s^{m-1} + \dots + a_0}{s^n + b_{n-1}s^{n-1} + \dots + b_0} \right]$$

$$= \alpha \frac{s^m}{s^n} \left[\frac{1 + \frac{a_{m-1}s^{m-1}}{s^m} + \dots + \frac{a_0}{s^m}}{1 + \frac{b_{n-1}s^{n-1}}{s^n} + \dots + \frac{b_0}{s^n}} \right]$$

$$\approx \frac{\alpha}{s^{n-m}} \quad \text{when } |s| \gg 1.$$

and thus, $\frac{\alpha}{s^{n-m}} \rightarrow 0$ as $s \rightarrow \infty$ also

and thus, there are q roots of the characteristic polynomial at $|s| = \infty$ in addition to m roots at the zeros of G for a total of n roots of the characteristic polynomial.

poles at infinity

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12:05 AM

In summary as $k \rightarrow \infty$
the roots of the characteristic
polynomial move toward the
 m zeros of G (roots of the polynomial
 $N(s)$) and $(n-m)=q$ values at $|\infty| = \infty$.

Typical Scenario

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Lets assume that G is stable transfer function.

Thus, d_G has roots only in the strict L.H.P. $\{s \mid \operatorname{Re}(s) < 0\}$.

⊗ Clearly when $k=0$, the characteristic polynomial has all roots stable [as the characteristic polynomial $n_k + d_G = d_G$] and the closed-loop system is stable.

⊗ Also, typically the transfer function G is such that, at least some roots of the characteristic polynomial move into the RHP. as k is increased

Roots on the imaginary axis

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Thus, the typical scenario is

- ① For K small the interconnection is stable and all closed-loop poles (the roots of the characteristic polynomial) are in the LHP
- ② For K very large at, least some closed-loop poles are located in the RHP
- ③ From continuity arguments it follows that there is a value of K for which

$$n(s)K + d_c(s) = 0$$
 for some values of s . Thus, the closed-loop poles cross the $j\omega$ axis as K is increased

from low values of K to large values of K , with an intermediate value of K ; say K_{cr} at which the characteristic polynomial has roots on the $j\omega$ axis.

Thus,

$K < K_{cr}$, the interconnection is stable

and

$K > K_{cr}$ the interconnection is unstable

with the characteristic polynomial having imaginary axis zeros for

$K = K_{cr}$.

The above condition is utilized by Bode plots to determine if $K < K_{cr}$ or $K > K_{cr}$

Critical Gain value

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Note that for imaginary axis zero of the characteristic polynomial there should be a frequency ω such that

$$K_{cr} N_G(j\omega) + d_G(j\omega) = 0$$

i.e.

$$\frac{K_{cr} N_G(j\omega)}{d_G(j\omega)} = -1$$

Thus,

$$K_{cr} G(j\omega) = -1$$

Thus,

$$|K_{cr}| |G(j\omega)| = 1$$

$$\text{and } \angle K_{cr} G(j\omega) = -180$$

$$\Rightarrow \angle G(j\omega) = -80.$$

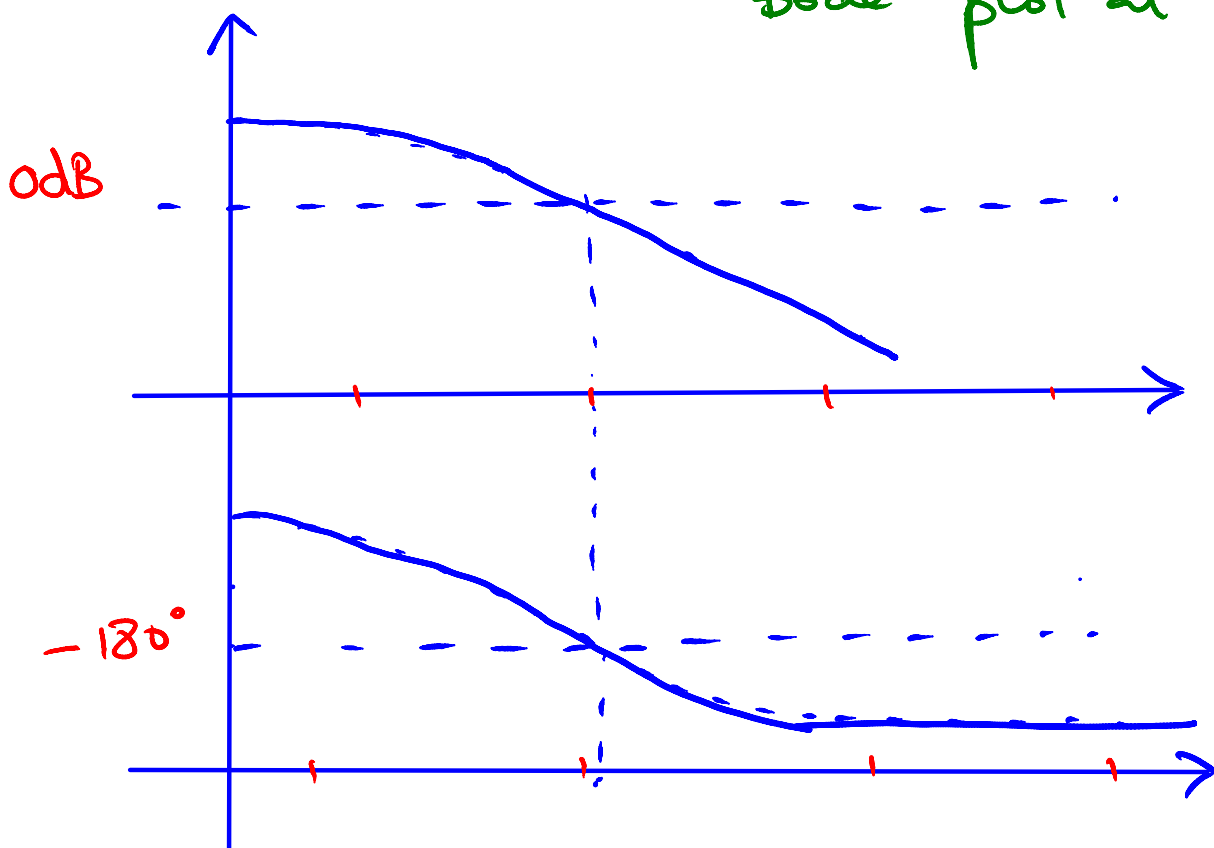
Thus, the Bode plot at K_{cr} of $K_{cr} G$ will look like

Critical Gain Value

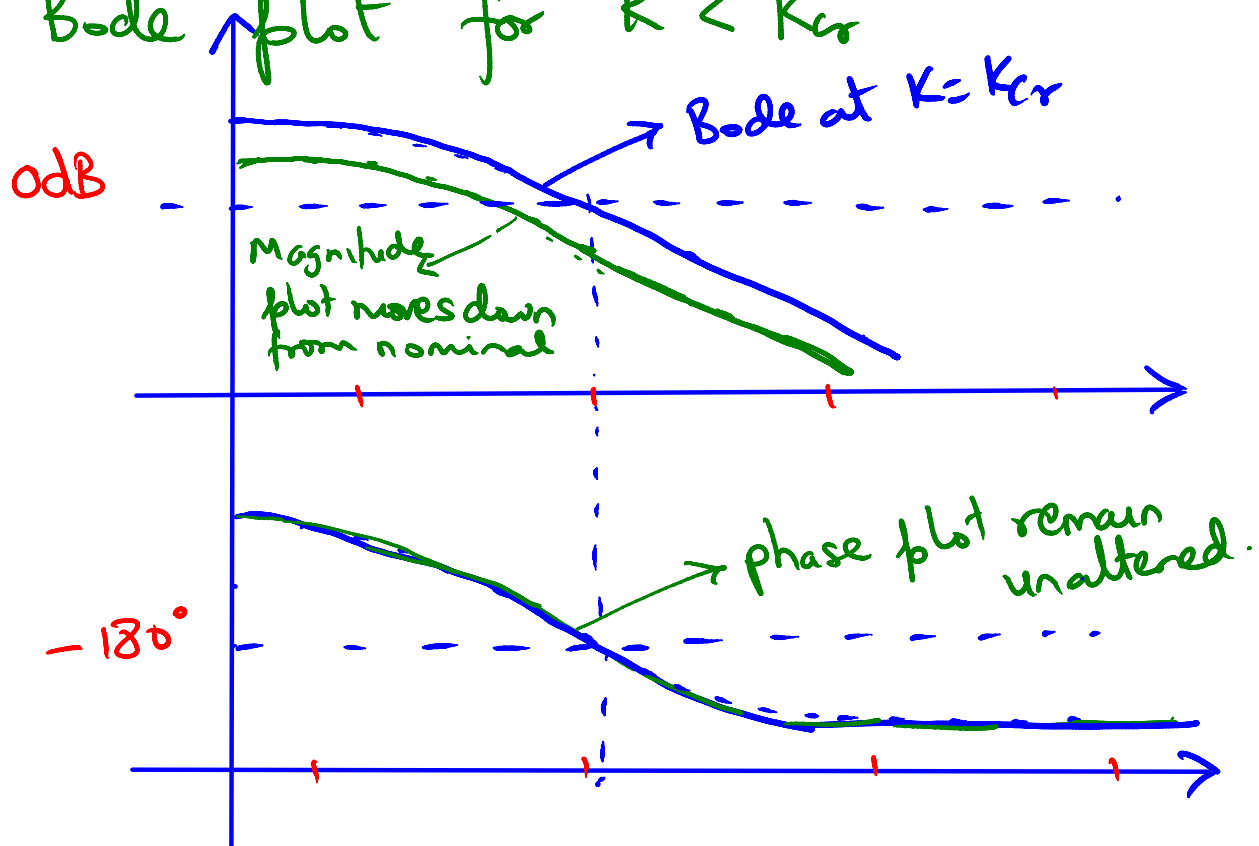
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Bode plot at K_{cr} .

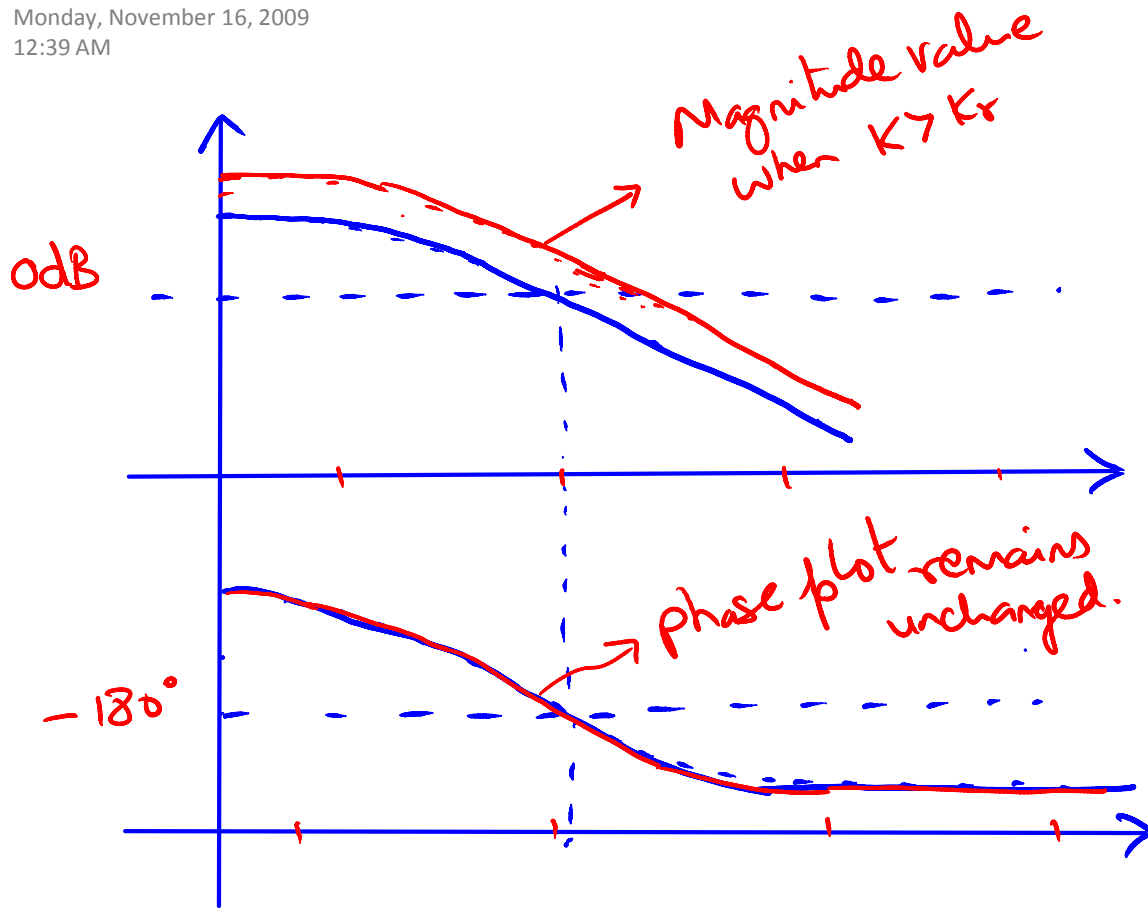


Bode plot for $K < K_{cr}$



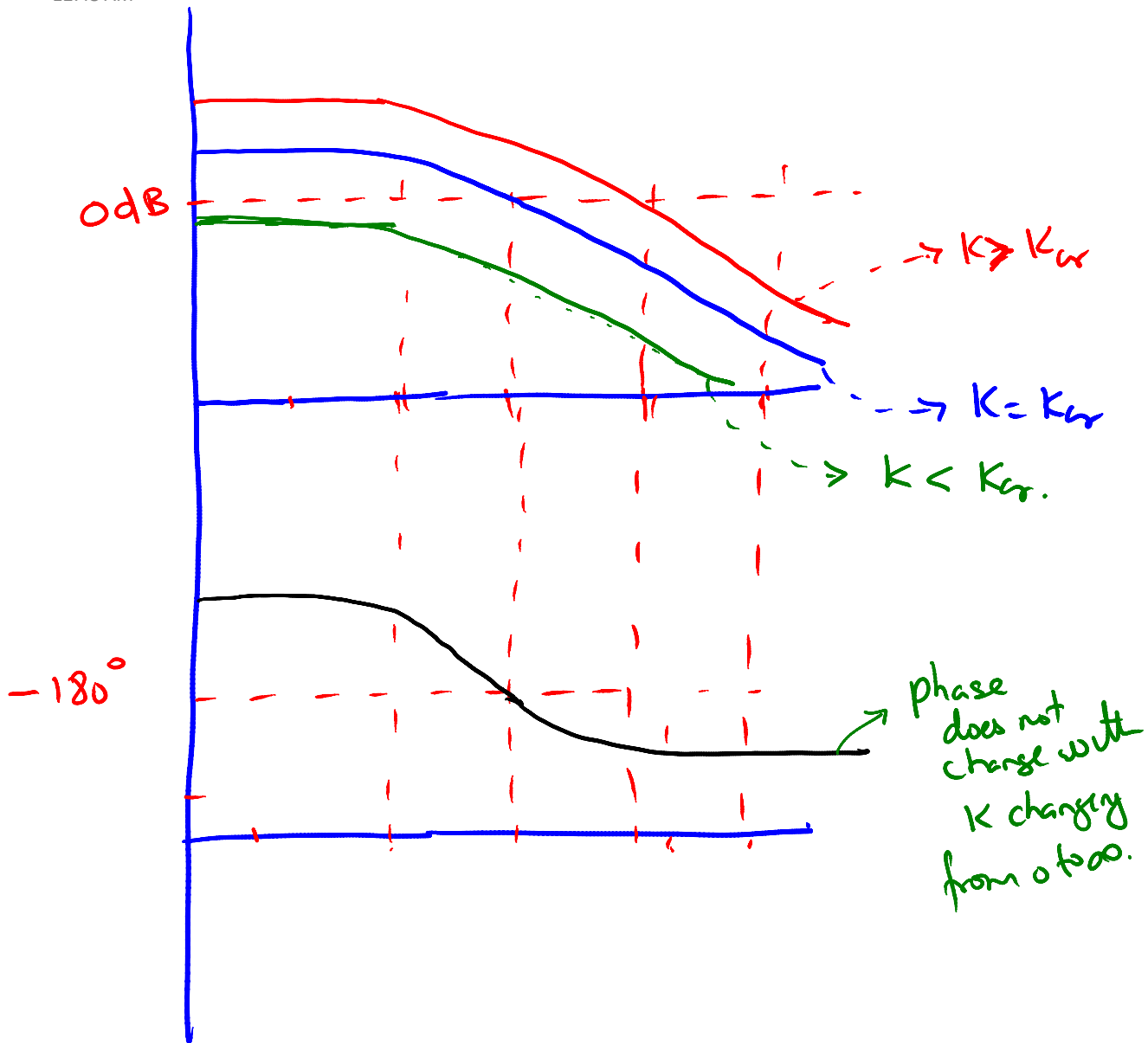
Value above critical value

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Summary Of a typical scenario

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- Note that when $K > K_{cr}$ then the typical interconnection is unstable
- Note that when $K < K_{cr}$ then the typical interconnection is stable.

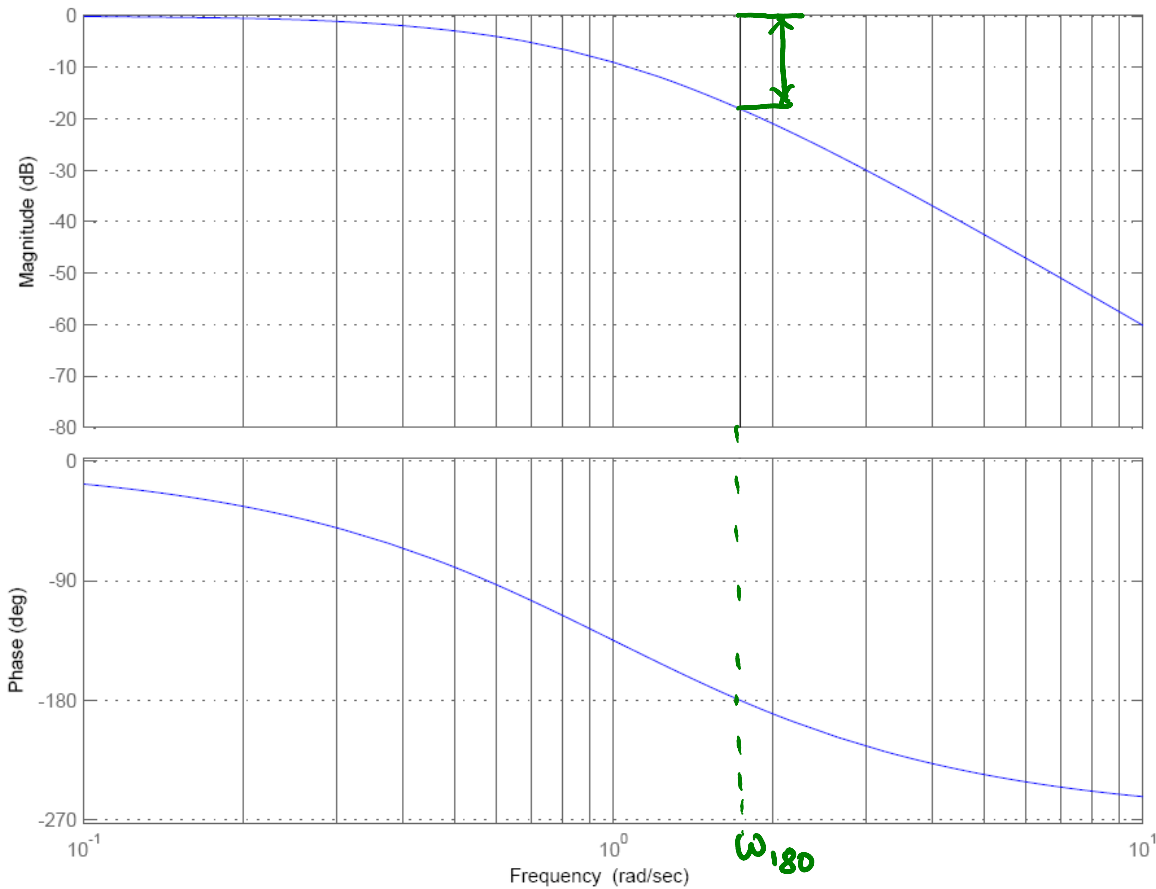
GAIN MARGIN

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How much can the gain be increased without rendering the unity negative feedback interconnection unstable is characterized by the Gain Margin

Bode Diagram
Gm = 18.1 dB (at 1.73 rad/sec), Pm = -180 deg (at 0 rad/sec)



Let ω_{180} be the frequency such that $\angle L(j\omega_{180}) = -180^\circ$. Then

$$\text{Gain Margin} = GM = -20 \lg_{10} |L(j\omega_{180})|$$

Gain Margin

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- ⊗ If the Gain Margin is positive a typical unity negative feedback interconnection is stable.

- ⊗ If the GM is negative then a typical unity negative feedback interconnection is unstable.

Phase Margin

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Note that the closed-loop system (the unity negative feedback interconnection) has a closed loop pole on the imaginary $j\omega$ axis if

$$1 + KG(j\omega) = 0 \quad \text{for some } \omega$$

i.e. when

$$1 + L(j\omega) = 0 \quad \text{for some } \omega$$

$$\Leftrightarrow \underline{\angle L(j\omega) = -180} \quad \text{for some } \omega.$$

$$|L(j\omega)| = 1$$

Let ω_{gc} be the value of ω such that (ω_{gc} is called the gain crossover frequency)

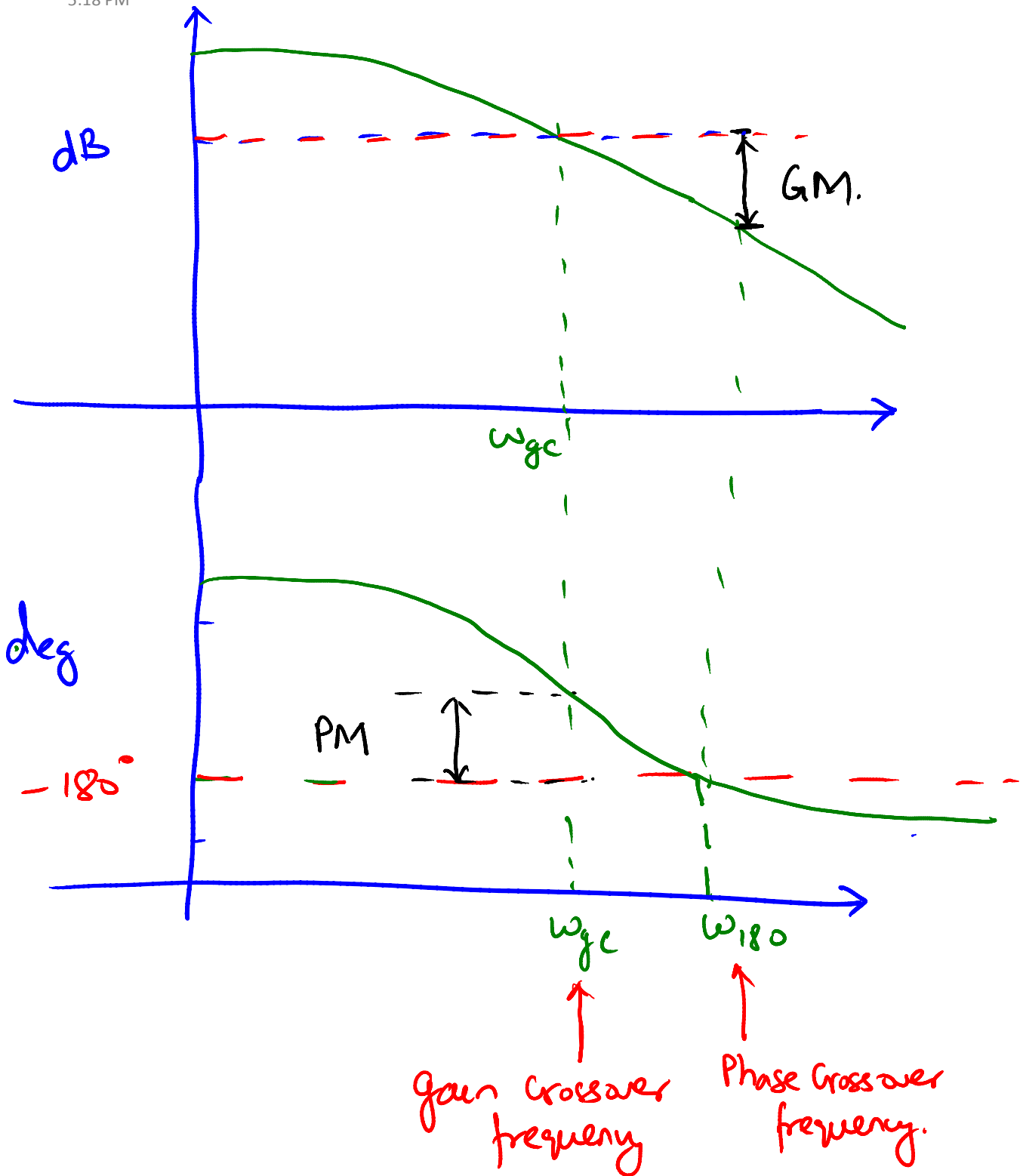
$$|L(j\omega_{gc})| = 1$$

and suppose the phase of L at ω_{gc} is $\underline{\angle L(j\omega_{gc})}$. Then

$$\boxed{PM = \underline{\angle L(j\omega_{gc})} - (-180) = 180 + \underline{\angle L(j\omega_{gc})}}$$

Phase Margin

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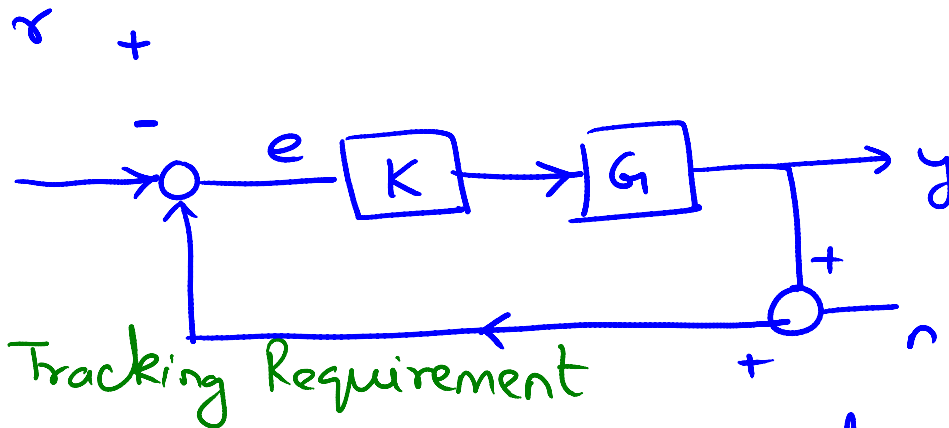
Phase Margin

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Thus, again (for a typical system)

- ⊕ If the phase Margin is positive
the closed-loop system is
stable
- ⊖ If the phase Margin is negative
the closed-loop system is unstable.

Consider the unity negative feedback closed-loop system



→ In many controls applications it is desired that the output y "tracks" r i.e. the error between y and r is required to be small.

→ Also, it is typical that the tracking is required not only for a specific input but

Frequency domain specifications

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over possibly many kinds of input with the characterization that the class of input signals to be tracked have bandwidth content within a certain frequency range ω_B .

Thus, if $r(t)$ is the reference trajectory to be tracked then

its Fourier transform

$$r(j\omega) = \int r(t) e^{-j\omega t} dt$$

is such that

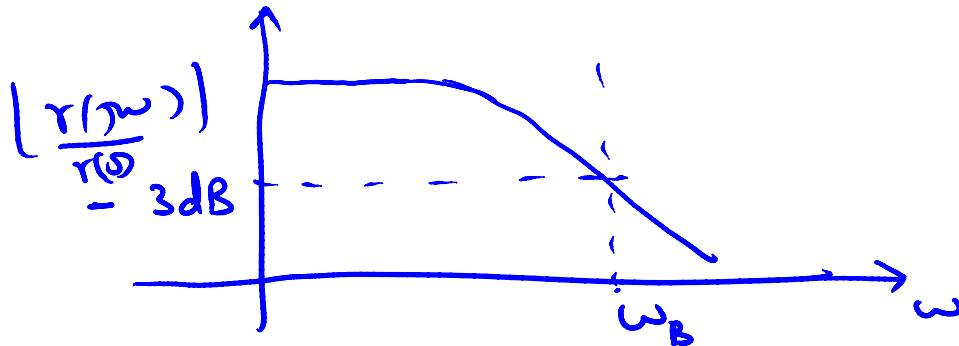
$$\left| \frac{r(j\omega)}{r(j0)} \right| \ll 1 \quad \forall \omega > \omega_B.$$

[note that we are using $r(t)$ and $r(j\omega)$ to represent the time domain and the frequency domain quantities respectively]

Tracking Requirement.

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Thus, it is known that $r(\omega)$ is such that



Thus, for $y(t)$ to track $r(t)$; we need the error $e(t) = r(t) - y(t)$ to be small. As $r(t)$ has a Fourier transform $r(\omega)$ with $| \frac{r(\omega)}{r(0)} |$ negligible beyond ω_B we need for good tracking $e(\omega)$ to be small for all $\omega \leq \omega_B$

Now

$$e(s) = \frac{1}{1 + G_1 K} r(s)$$

Tracking Requirement

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and thus, we need

$$\frac{e(s\omega)}{r(s\omega)} = \frac{1}{1 + G(s\omega)K(s\omega)}$$

to be small for all $\omega \leq \omega_B$.

Sensitivity Transfer function:

The transfer function

$$S = (1 + GK)^{-1}$$
 is called

the sensitivity transfer function.

Thus, the tracking requirement translates to having

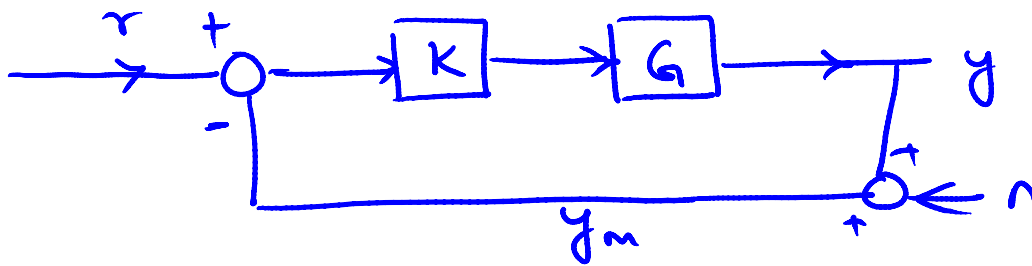
$$|S(s\omega)| \text{ to be small}$$

for all $\omega \leq \omega_B$.

ω_B is called the tracking bandwidth.

Noise Rejection

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Any measurement procedure is not precise; every sensor introduces noise, n . However, it is not desired that the noise n affect the output y .

The transfer function between noise n to the output y is given

$$\frac{-GK}{1+GK}$$

and thus, it is typically desired that this transfer function is small beyond a

Noise Rejection

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frequency ω_{BT} .

Complimentary Sensitivity transfer function.

$$T = \frac{GK}{1+GK}$$

is called the complimentary transfer function.

Thus, we need

$|T(\omega)|$ to be small
beyond $\omega > \omega_{BT}$.

Summary

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In summary for

(a) Tracking we need

$$S = \frac{1}{1+GK} = \frac{1}{1+L} \text{ to}$$

be small in the frequency

range

$$\omega \leq \omega_B$$

(b) For noise rejection we need

$$T = \frac{GK}{1+GK} = \frac{L}{1+L} \text{ to}$$

be small in the frequency

range

$$\omega \gg \omega_{BT}$$

A fundamental limitation

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Note that

$$S(\omega) = \frac{1}{1+L(\omega)}$$

$$\text{and } T(\omega) = \frac{L(\omega)}{1+L(\omega)}$$

and therefore

$$\begin{aligned} S(\omega) + T(\omega) &= \frac{1}{1+L(\omega)} + \frac{L(\omega)}{1+L(\omega)} \\ &= \frac{1+L(\omega)}{1+L(\omega)} \\ &= 1. \end{aligned}$$

and thus, at the same frequency ω it is not possible to have both $S(\omega)$ and $T(\omega)$ small. Fortunately

the requirements on S and T are possible on different frequency ranges $[0, \omega_B)$ and (ω_B, ∞) respectively.

Translating closed-loop objectives to open loop tf

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Here we will translate the conditions on the closed loop map to the open-loop transfer function L .

Note that

we want

$|S(j\omega)| = \left| \frac{1}{1+L(j\omega)} \right|$ to be small in the frequency region $\omega \in [0, \omega_B]$

This implies we need

$|L(j\omega)|$ to be large in the frequency region $[0, \omega_B]$

Typically, we need

$$20 \log_{10} |L(j\omega)| \geq +3\text{dB} \quad \text{for } \omega \in [0, \omega_B]$$

Closed-loop objective to open-loop tf.

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The noise-rejection objective is that the complementary transfer function

$$|T(j\omega)| = \left| \frac{L(j\omega)}{1 + L(j\omega)} \right|$$

to be small in $\omega \in [\omega_{BT}, \infty)$.

i.e.

$$|T(j\omega)| = \left| \frac{1}{\frac{1}{L(j\omega)} + 1} \right| \text{ to}$$

be small in $\omega \in [\omega_{BT}, \infty)$

Thus, we require $|L(j\omega)|$ to be small in the range $\omega \in [\omega_{BT}, \infty)$.

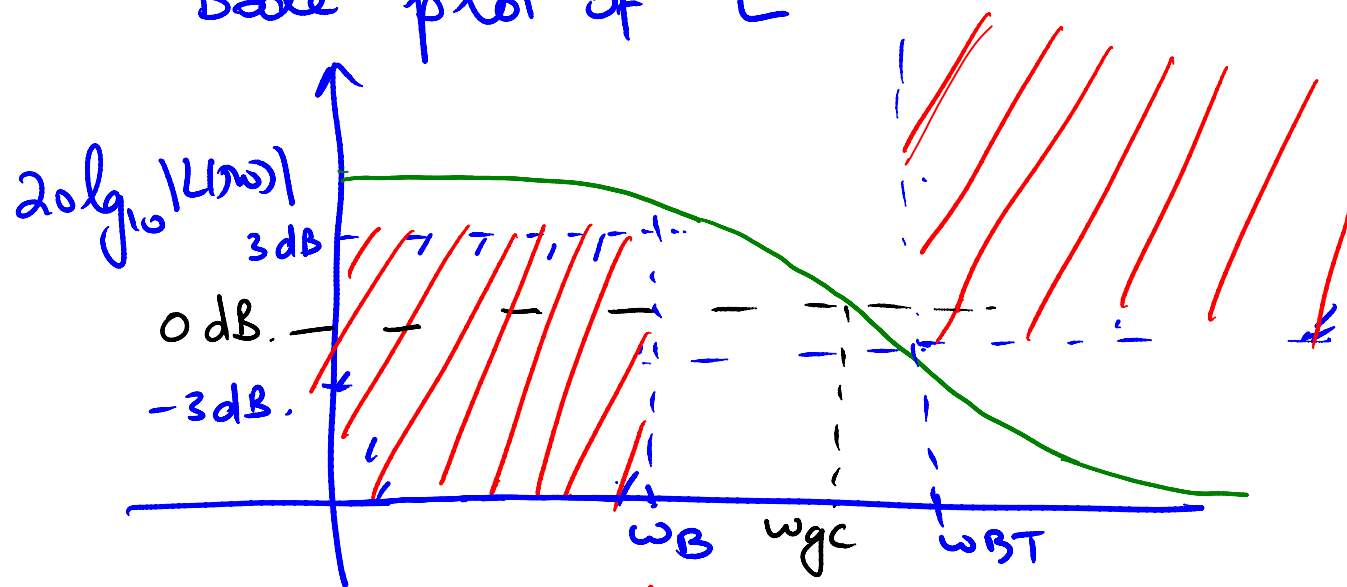
Typically it is needed that

$$20 \log_{10} |L(j\omega)| \leq -3 \text{ dB for } \omega \gg \omega_{BT}.$$

Specifications on the open-loop tf.

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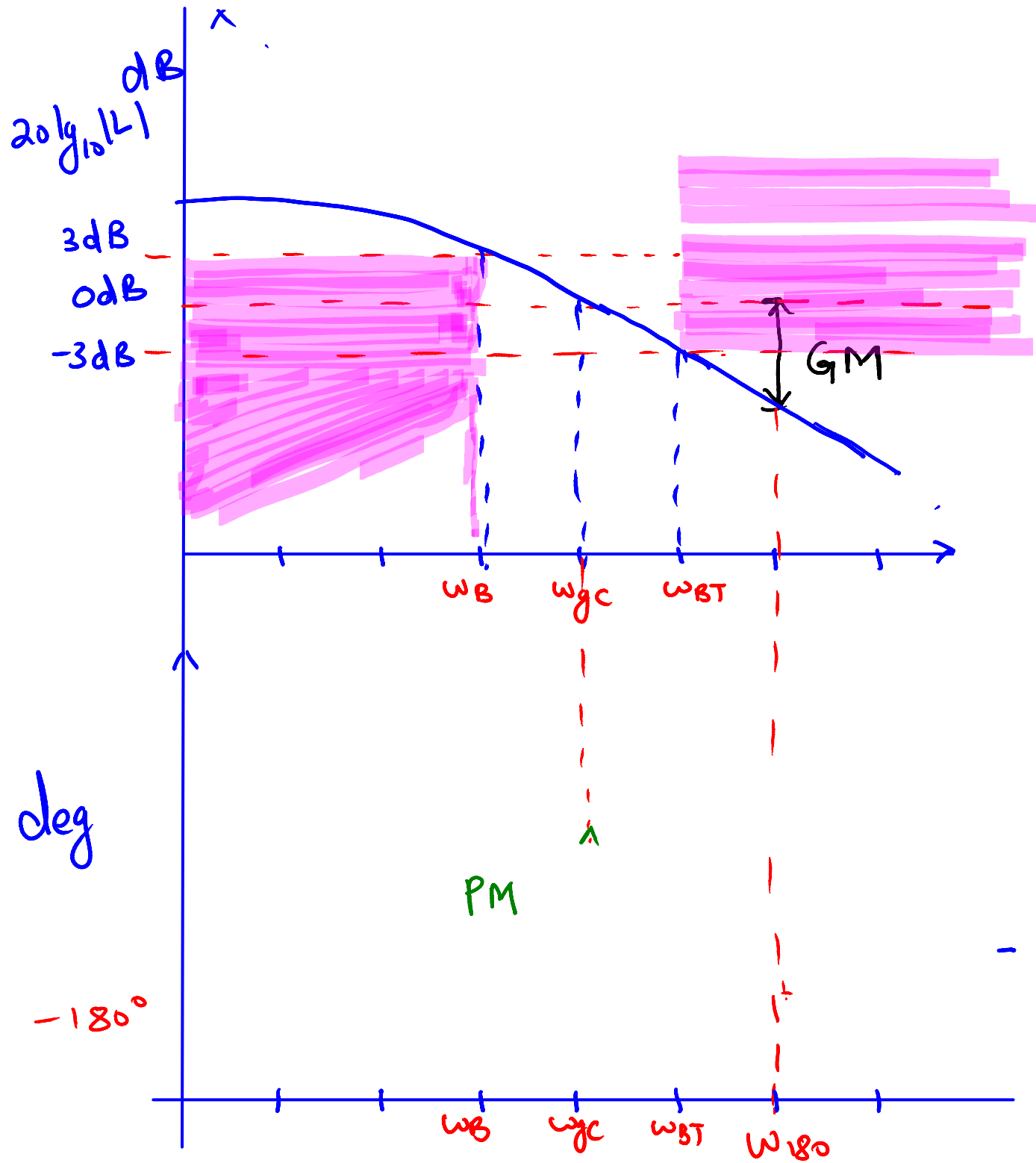
In Summary we need the following magnitude part of the Bode plot of L



→ The red region has to be avoided by the transfer function $L(s)$.

Summary

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Time-domain Specifications

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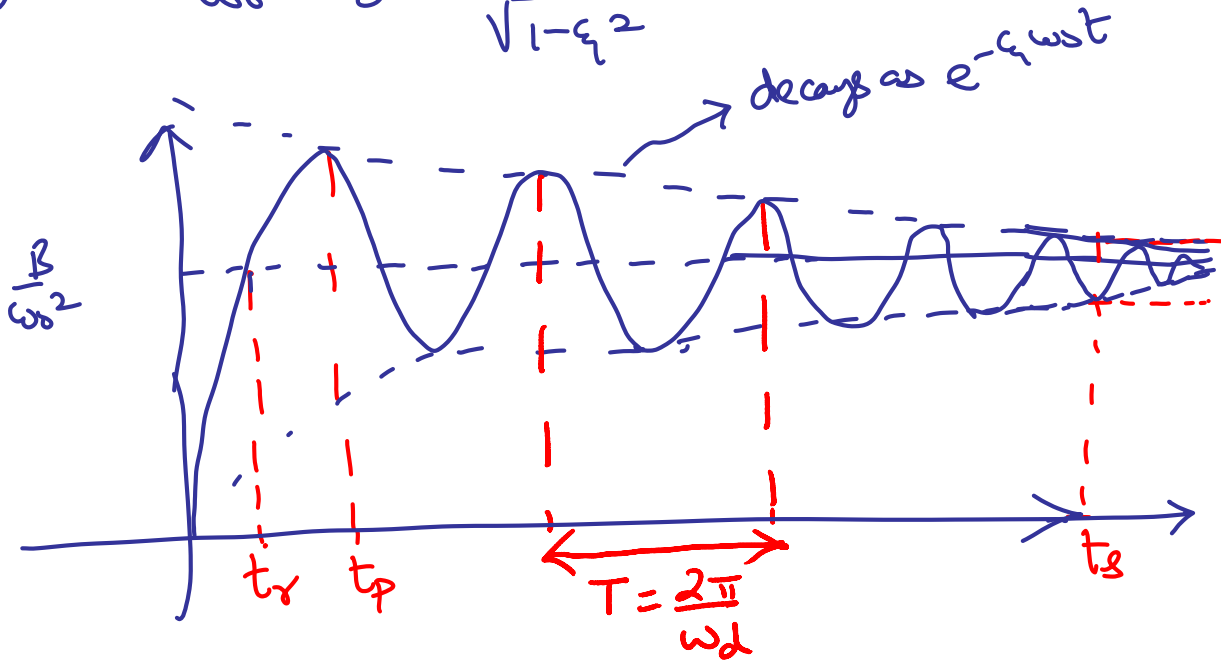
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Plots of Step response of ϕ

$$\frac{B}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

given by

$$y(t) = \frac{B}{\omega_n^2} \left[1 - \frac{e^{-\zeta\omega_n t}}{\sqrt{1-\zeta^2}} \sin(\omega_d t + \phi) \right]$$



- ① t_r = rise time: the first time the output reaches the steady state value $y_s = B/\omega_n^2$
- t_p = peak time; the time the maximum is reached
- t_s = settling time: the time after which output remains within 2% of the steady state y_s
- y_s = steady state value B/ω_n^2
- $M_p = \frac{y(t_p) - y_s}{y_s}$

Time domain Specifications

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$$\rightarrow y_{ss} = B/\omega_0^2$$

$$\rightarrow t_p = \frac{\pi}{\omega_d} ; \quad \omega_d = \omega_0 \sqrt{1-\zeta^2}$$

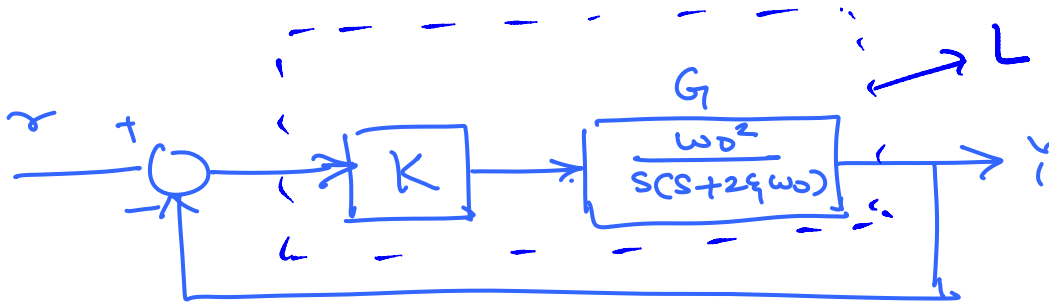
$$\rightarrow t_r = \frac{\pi - \phi}{\omega_d} ; \quad \phi = \cos^{-1} \zeta.$$

$$\rightarrow t_s = \frac{4}{\zeta \omega_0} ; \quad 2\% \text{ settling time}$$

$$\rightarrow M_p = e^{(-\pi \zeta / \sqrt{1-\zeta^2})}$$

Second order prototype

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→ open-loop transfer function

$$L = \frac{K\omega_0^2}{s^2 + 2\zeta\omega_0 s}$$

→ closed-loop transfer function from $r \mapsto y$ is

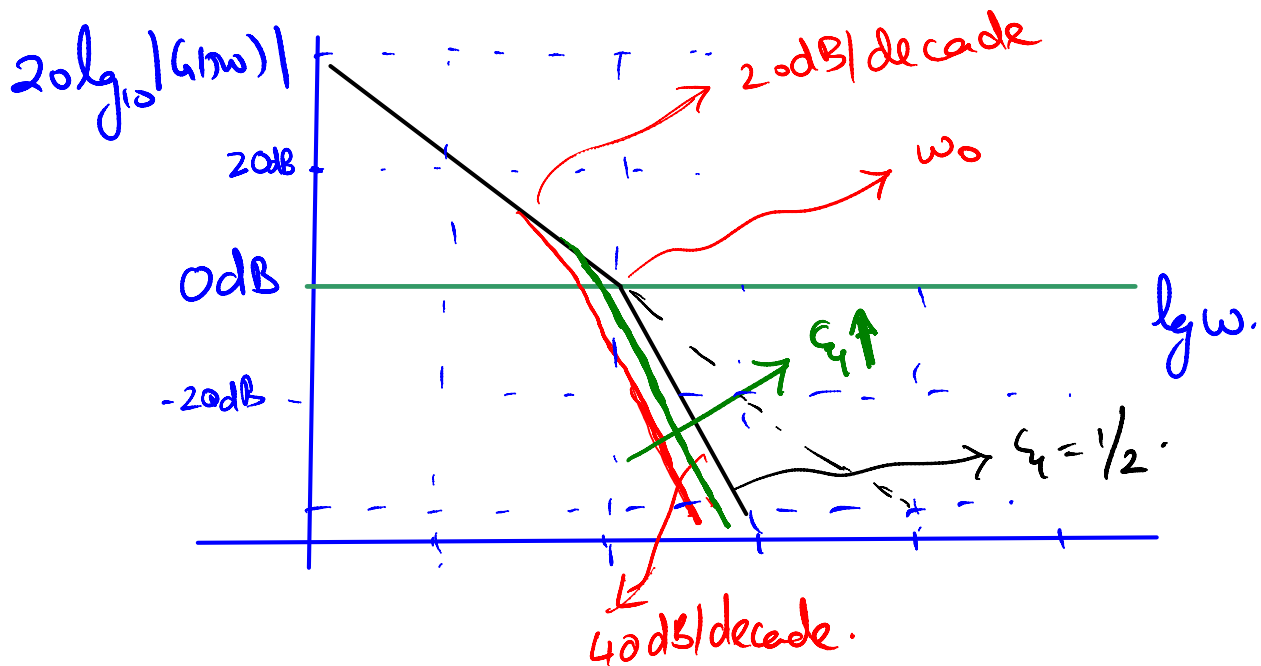
$$\frac{L}{1+L} = \frac{K\omega_0^2}{s^2 + 2\zeta\omega_0 s + K\omega_0^2}$$

→ for $K=1$ we have the closed-loop $r \mapsto y$ have the prototype second-order transfer function

$$\frac{\omega_0^2}{s^2 + 2\zeta\omega_0 s + \omega_0^2}$$

Magnitude of the open-loop L

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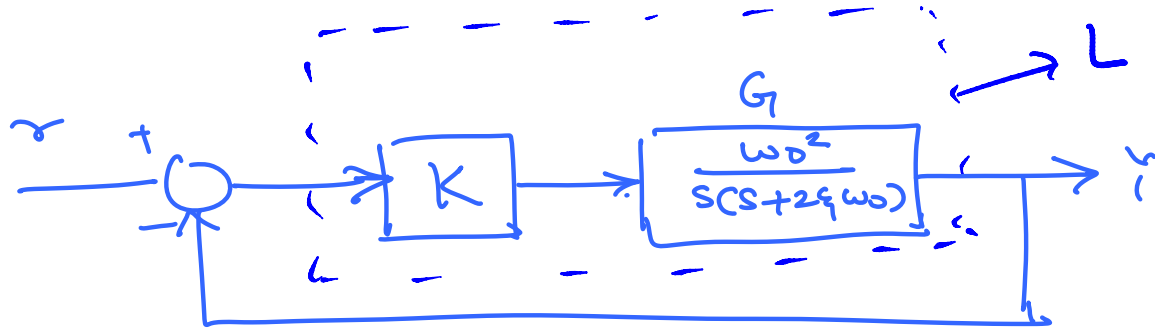
→ A good thumb rule is that the gain cross-over frequency is

$$\omega_{gc} = \frac{\omega_0}{1.4}$$

— A good thumb rule is that the closed-loop bandwidth is $(1.2 - 1.6) \omega_{gc}$.

Phase margin of second order prototype

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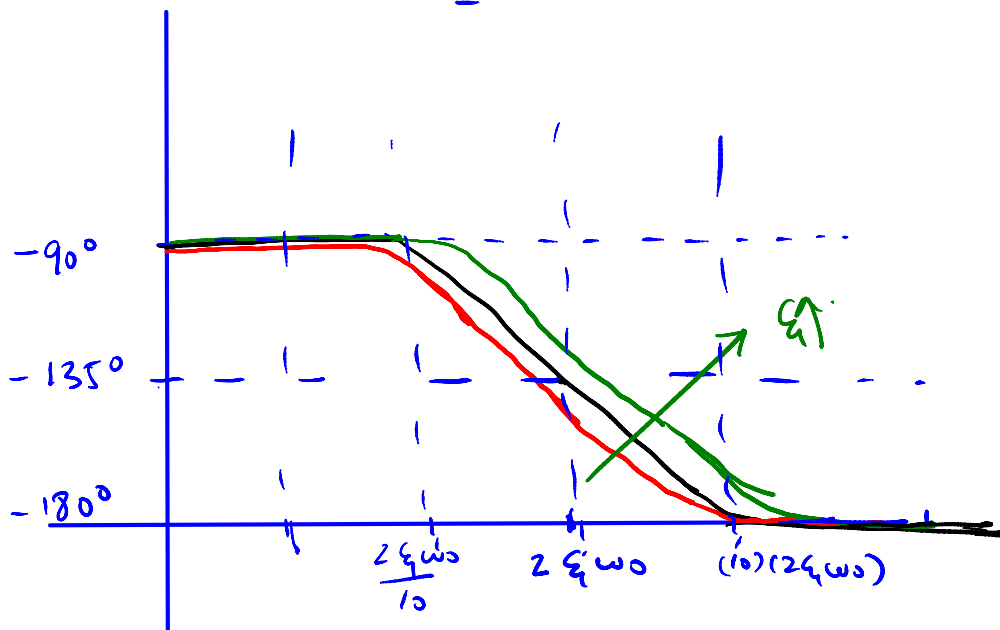


→ The open-loop transfer function

$$L = \frac{k\omega_0^2}{s^2 + 2\zeta\omega_0 s}$$

$$\underline{L}(j\omega) = \frac{\omega_0}{j\omega (j\omega + 2\zeta\omega_0)}$$

$$= -\pi/2 - \text{atan}\left(\frac{\omega}{2\zeta\omega_0}\right)$$



Phase Margin of a Second order prototype

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$$\text{Phase margin} = \pi + \angle G(j\omega_c)$$

$$= \pi - \frac{\pi}{2} - \text{atan}\left(\frac{\omega_c c}{2\zeta\omega_n}\right)$$

$$= \frac{\pi}{2} - \text{atan}\left[\frac{(\sqrt{1+4\zeta^4}-2\zeta^2)^{1/2}}{2\zeta}\right]$$

$$\zeta = 0.5 ;$$

$$PM = 51^\circ$$

$$\zeta = 0.6 ;$$

$$PM = 59^\circ$$

$$\zeta = 0.7 ;$$

$$PM = 65^\circ$$

$$\zeta = 1$$

$$PM = 70^\circ$$

For most typical systems

$$PM = 100\zeta \text{ in degrees.}$$

Time-domain Specs. and Bode plot.

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The time domain Specs. depend on ζ and ω_0

$$\rightarrow t_p = \frac{\pi}{\omega_d} ; \quad \omega_d = \omega_0 \sqrt{1-\zeta^2}$$

$$\rightarrow t_r = \frac{\pi - \phi}{\omega_d} ; \quad \phi = \tan^{-1} \zeta$$

$$\rightarrow t_s = \frac{4}{\zeta \omega_0} ; \quad 2\% \text{ settling time}$$

$$\rightarrow M_p = e^{(-\pi \zeta / \sqrt{1-\zeta^2})}$$

$$\rightarrow \omega_{gc} \cong \frac{\omega_0}{1.4} ; \quad PM \cong 180 + \underline{L(j\omega_{gc})} \cong 100\zeta$$

→ ω_{gc} and PM can be determined by the Bode plot; thus fixing ω_0 and ζ .

→ Note that the steady state error due to steps, ramps or parabolic inputs gets determined by the behaviour of $L(j\omega)$ for ω close to zero (see next page).

→ ζ and ω_0 all get approximately determined by the Bode plot and thus determine time-domain properties.

Selection of type

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⊗ Note that the steady state error in tracking steps is

$$\frac{1}{1+K_p}$$

where $K_p = G(0)K(0) = L(0)$

* Steady state error tracking ramps is

$$\frac{1}{K_v}$$

where $K_v = \lim_{s \rightarrow 0} s G(s)K(s)$
 $= \lim_{s \rightarrow 0} s L(s)$

⊗ Steady state error for tracking parabolic inputs is

$$\frac{1}{K_a}$$

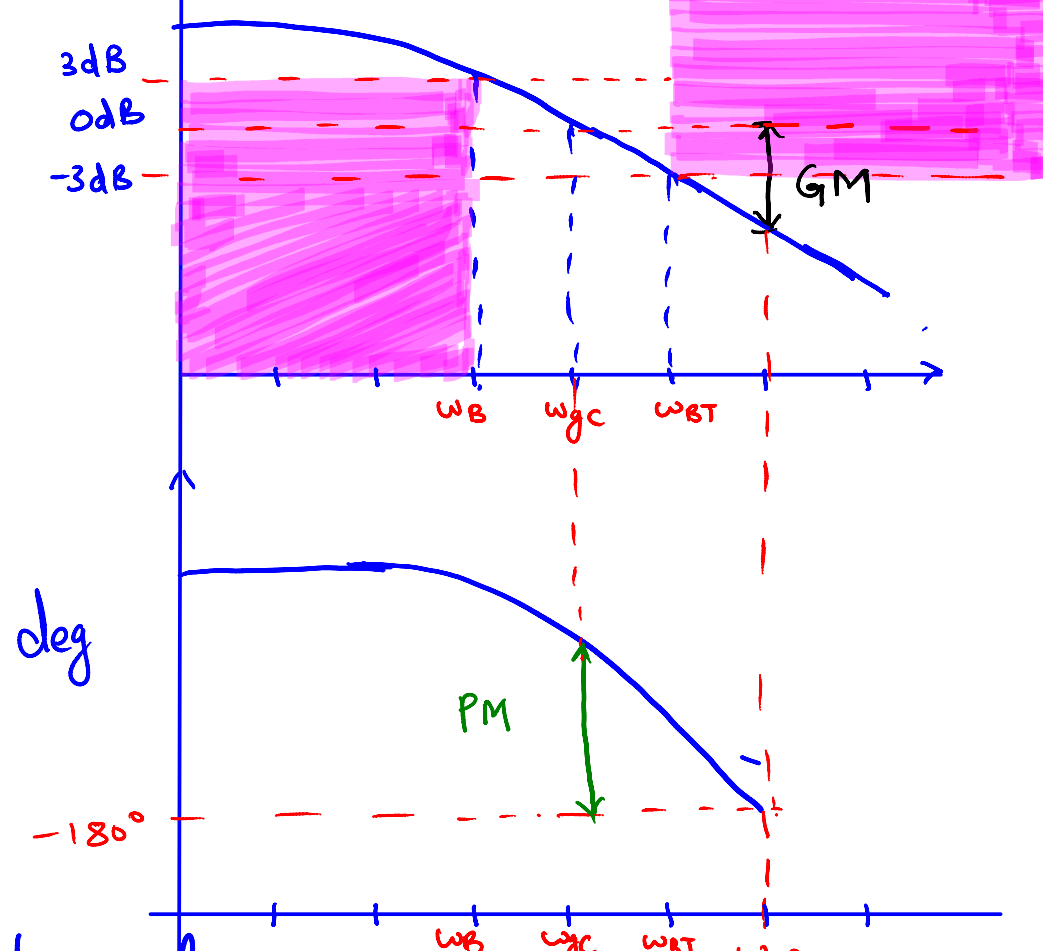
where $K_a = \lim_{s \rightarrow 0} s^2 G(s)K(s)$
 $= \lim_{s \rightarrow 0} s^2 L(s)$

⊗ Therefore need -20dB/decade slope near DC for tracking steps (for the Bode plot)

⊗ Need -40dB/decade slope near DC for tracking ramps

⊗ Need -60dB/decade slope near DC for tracking parabolic inputs

Summary



closed-loop Specs:

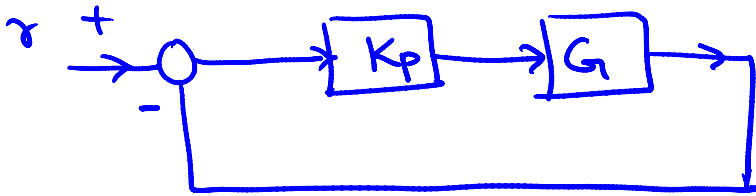
OPEN LOOP

- ⊕ Bandwidth BW = $(1.2 - 1.6) \omega_{gc}$
- ⊕ PM (in degrees) = 100%
Can be read as shown above
- ⊕ GM
Can be read as shown above.
- ⊗ $M_p = e^{-\pi \zeta / \sqrt{1 - \zeta^2}}$
 $\zeta = (PM \text{ in deg}) / 100$
- ⊗ $t_s = 4 / \zeta \omega_0$; $t_p = \pi / (\omega_0 \sqrt{1 - \zeta^2})$
 $\omega_0 = 1.4 \omega_{gc}$
- ⊕ $t_r = (\pi - \cos^{-1} \zeta) / (\omega_0 \sqrt{1 - \zeta^2})$
- ⊕ Steady state behavior
determined by the behavior of $L(s)$ near $s=0$.

Controller design

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Proportional controller:

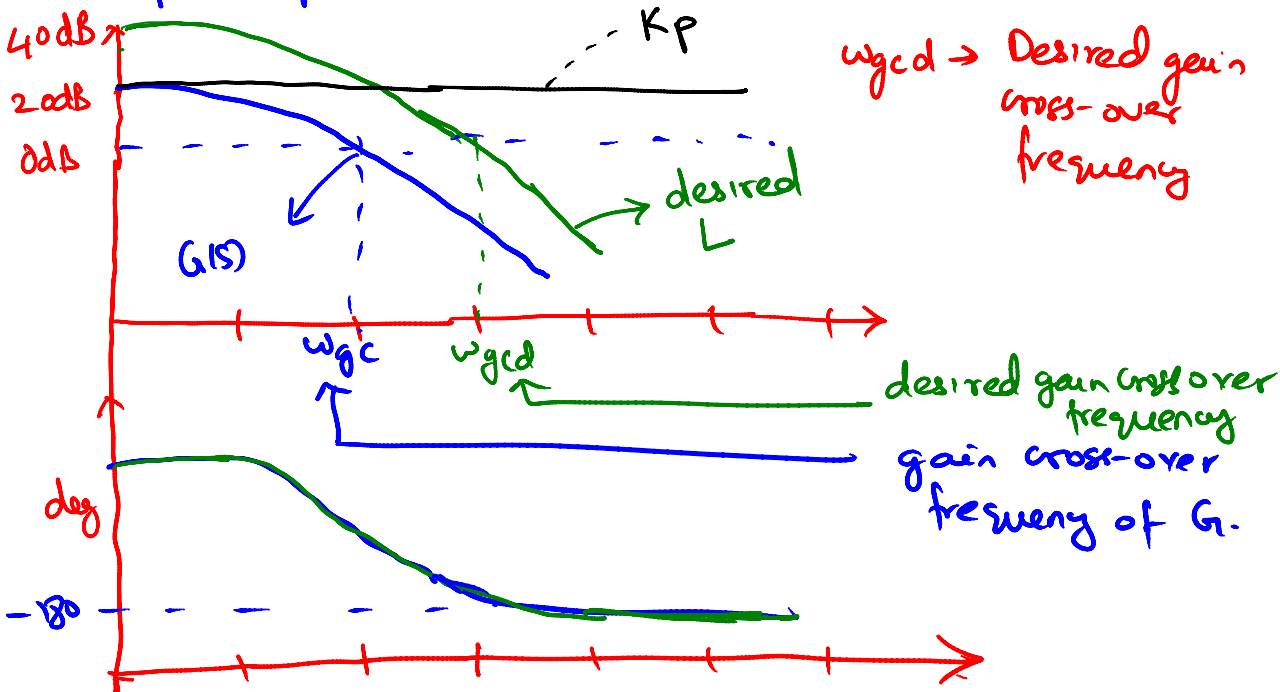


① $L = K_p G_1$.

② proportional controller $K = K_p$ a positive constant.

→ only one parameter to determine K_p .

Note that K_p can only move the magnitude Bode plot up or down without affecting the phase plot



Proportional controller.

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- ① Note that desired gain crossover frequency is ω_{gcd}
- ② To move $G(s)$ to desired L multiply by K_p i.e. $20 \lg_{10} K_p = 20 \text{ dB}$

$$\Rightarrow \lg_{10} K_p = 1$$

$$\Rightarrow K_p = 10.$$

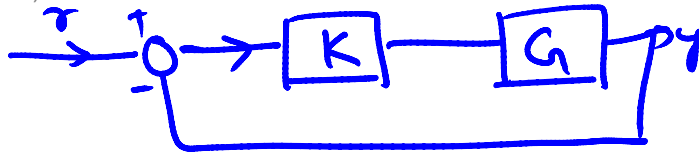
If the need is to have only one closed-loop characteristic chosen K_p can be similarly utilized

There is only one parameter to be designed!

- choose K to satisfy e_{ss} (steady state error)
- choose K to satisfy PM requirements
- choose K to satisfy t_r, t_s, M_p requirements.

Proportional integral controller

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$$\rightarrow K = K_p + \frac{K_I}{s} = \frac{sK_p + K_I}{s}$$

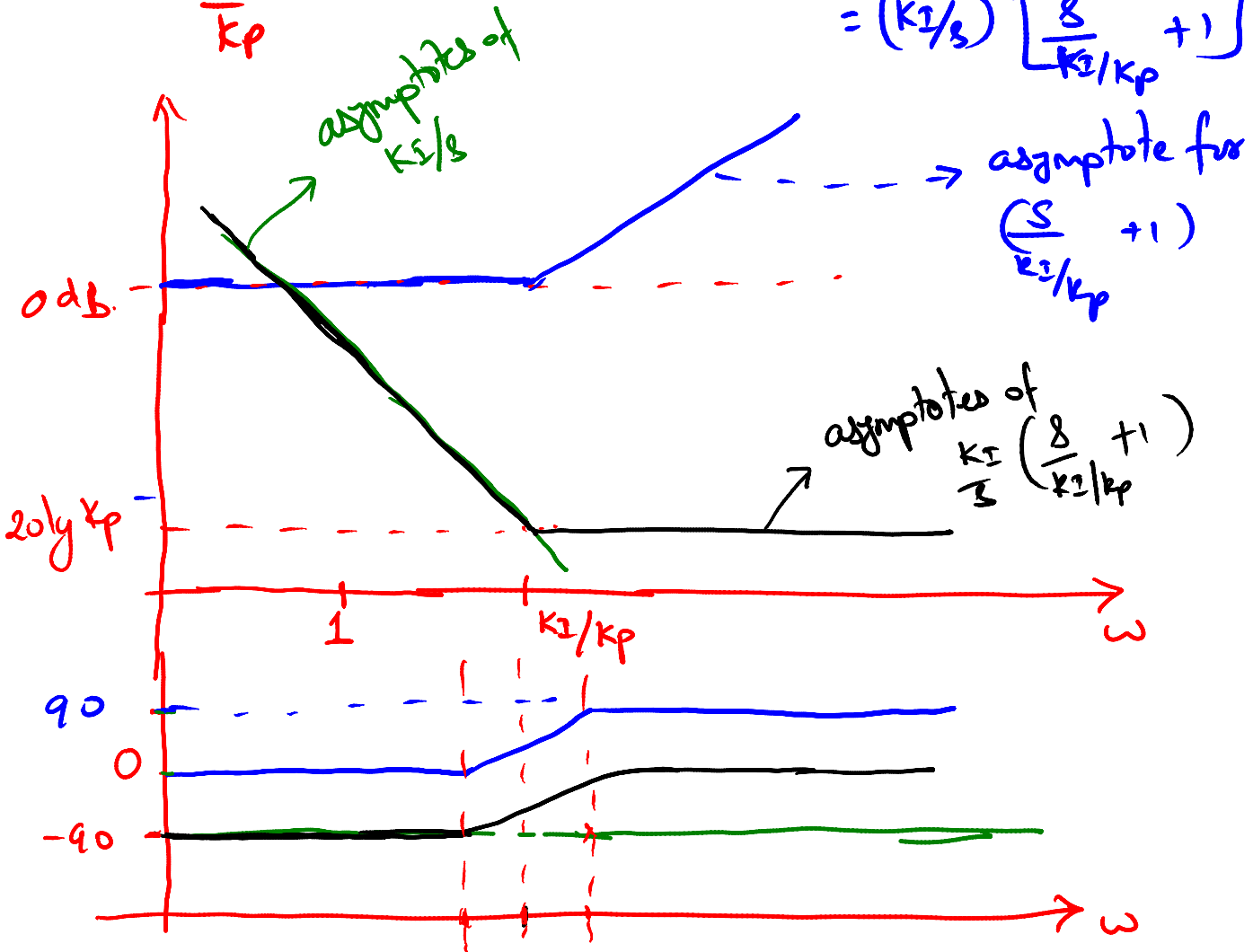
⊕ K_p is the proportional gain

⊕ K_I is the integral gain

→ In the standard form $K(s) = \frac{K_I}{s} \left[\frac{sK_p}{K_I} + 1 \right]$

Assume $\frac{K_I}{K_p} > 1$.

$$= (K_I/s) \left[\frac{s}{K_I/K_p} + 1 \right]$$



Proportional-integral controller

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→ Note that the phase provided by a P-I controller is always negative.

⊗ This will worsen the Phase-margin

⊗ Phase ≈ 0 for $\omega \gg K_I/K_P$

→ The P-I controller has a slope of -20 dB/decade in the dc region

⊗ This will increase the type of the system.

→ Suppose G is Type 0 then with PI controller L will have type 1

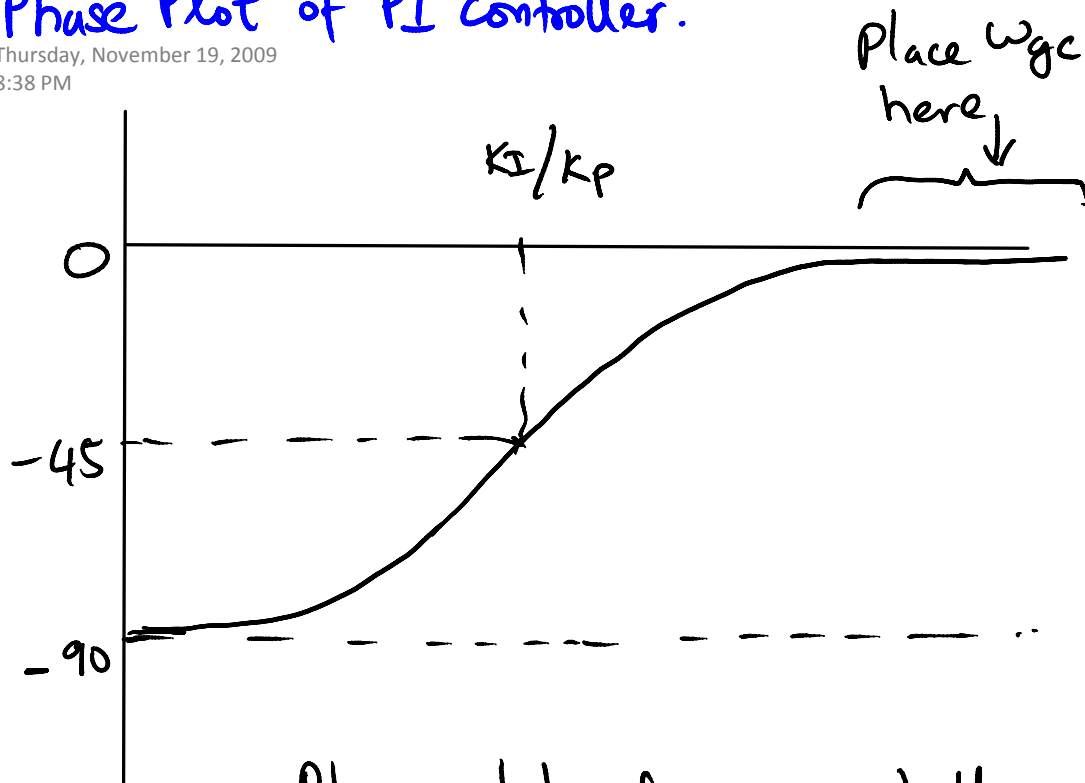
Thus the PI controller increases type of the interconnection

⊗ Phase-margin can only be reduced by a PI controller; a destabilizing effect.

→ To reduce the destabilizing effect choose K_I/K_P such that $\omega_{gc} \gg K_I/K_P$ so that negative phase contribution due to PI controller is minimal effect on phase margin.

Phase Plot of PI controller.

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Phase-plot of PI controller.

→ choose $\frac{K_I}{K_P}$ such that $\omega_{gc} = \alpha \frac{K_I}{K_P}$
where $5 \leq \alpha \leq 10$.

Guidelines for designing PI controllers

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Design 1:

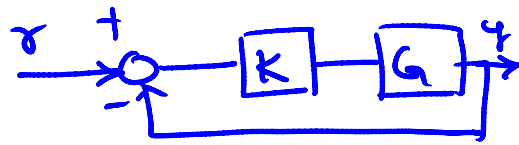
- ① Design K_p to fix M_p , phase-margin or ...
- ② Choose K_I such that
$$\frac{K_I}{K_p} \leq \frac{\omega_{gcd}}{\alpha}; \quad \alpha \text{ is between } 5 \text{ and } 10$$
and $\frac{K_I}{s} G(s)$ satisfy steady state error requirement.

Design 2:

- ① Select K_I such that
$$K_v = \lim_{s \rightarrow 0} \left[s \frac{K_I}{s} G(s) \right]$$
to meet steady state error due to ramps specification
- ② Note that the characteristic equation is
$$K_p G(s) + \frac{K_I}{s} G(s) + 1 = 0$$
$$\Rightarrow K_p \frac{G(s)}{\frac{K_I}{s} G(s) + 1} + 1 = 0$$
- ③ Do proportional design for $\frac{G(s)}{\frac{K_I}{s} G(s) + 1}$

PI design example

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Suppose $G(s) = \frac{500}{s^2 + 6s + 5}$

Design a PI controller to meet the following specifications on the closed-loop tf between r and y

(a) $M_p \leq 16\%$

(b) e_{ss} (steady state error) due to ramp input ≤ 0.1

Solution:

$$\text{Desired } M_p = e^{-\frac{\pi \zeta}{\sqrt{1-\zeta^2}}} \leq 0.16$$

$$\Rightarrow -\frac{\pi \zeta}{\sqrt{1-\zeta^2}} \leq \ln 0.16.$$

$$\Rightarrow \frac{\pi \zeta}{\sqrt{1-\zeta^2}} \geq -\ln 0.16$$

$$\Rightarrow \frac{\pi \zeta}{\sqrt{1-\zeta^2}} \geq +1.8326$$

$$\Rightarrow \zeta \geq 0.5039.$$

\therefore The Phase Margin desired

$$PM_d = 100\zeta \approx 50 + 7 \text{ (Safety margin)} \\ = 57 \text{ degrees.}$$

Steady state error

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→ The transfer function between r and error e is

$$\frac{e}{r} = \frac{1}{1+L}$$

$$\therefore e(s) = \frac{1}{1+L(s)} r(s)$$

when r is a ramp then; $e(s) = \frac{1}{1+L} \cdot \frac{1}{s^2}$

and the final value theorem states that

$$\lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} s e(s)$$

$$= \lim_{s \rightarrow 0} \left[\frac{1}{1+L(s)} \cdot \frac{s}{s^2} \right]$$

$$= \lim_{s \rightarrow 0} \frac{1}{sL(s)+s} = \frac{1}{\lim_{s \rightarrow 0} sL(s)} = \frac{1}{K_v}$$

\therefore For $e_{ss} \leq 0.1$ we need

$$\frac{1}{K_v} \leq 0.1 \Rightarrow K_v \geq 10.$$

$$K_v = \lim_{s \rightarrow 0} sL(s) = \lim_{s \rightarrow 0} s \left[K_p + \frac{K_I}{s} \right] G(s)$$

$$= \lim_{s \rightarrow 0} (sK_p L(s) + K_I L(s)) = K_I G(0)$$

[assuming stability of the interconnection]

\therefore we need $K_I G(0) \geq 10.$

Example Continued.

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Thus, the specifications on the closed-loop translate to

(a) Phase Margin $\approx 57^\circ$

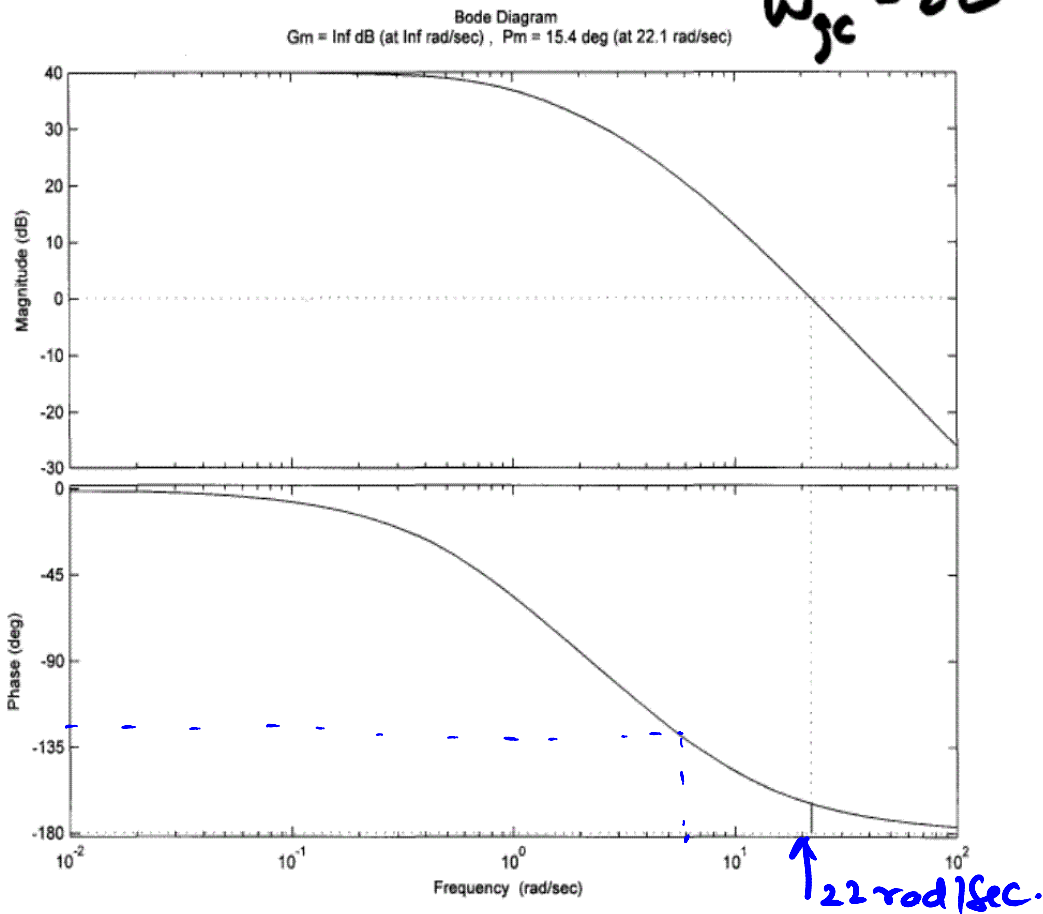
(b) $K_I G(0) \geq 10$.

$$\Rightarrow K_I 100 \geq 10 \Rightarrow K_I \geq \frac{1}{10} = 0.1$$

Bode of G_r

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$PM = 15^\circ$
 $\omega_{gc} = 22$



Commentary: (i) The cross-over frequency with controller = 1; is $\omega_{gc} = 22$ rad/sec.

(*) The associated phase Margin $PM_{existing} = 15^\circ$.

(ii) Desired phase Margin $PM_d = 57^\circ$

∴ Extra phase needed

(*) Note that at $\omega = 4.39$ $|G(\omega)| \approx -118$

∴ If gain crossover is moved to $\omega = 4.39$
PM will be $180 - 118 = 62$ degrees.

Fixing K_p and K_I

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→ Choose K_p so that

the gain crossover frequency of

$L_1 = K_p G(s)$ is at

$$\omega_{gcd} = 4.39 \text{ rad/sec}$$

$$\therefore \text{let } K_p = \frac{1}{|G(j4.39)|} \cong 0.06$$

$$\left[\text{This will result in } |L_1(j4.39)| = |K_p G(j4.39)| = 1. \right]$$

→ Now choose K_I such that nothing gets effected near gain crossover frequency. This can be achieved by choosing $\frac{K_I}{K_p} = \frac{\omega_{gcd}}{\alpha}$; α in the range 5-10

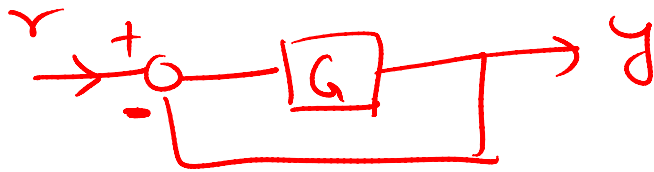
$$= \frac{(4.39)}{\alpha} 0.06$$

→ Choosing $\alpha = 6$ we have $K_I = 0.044$

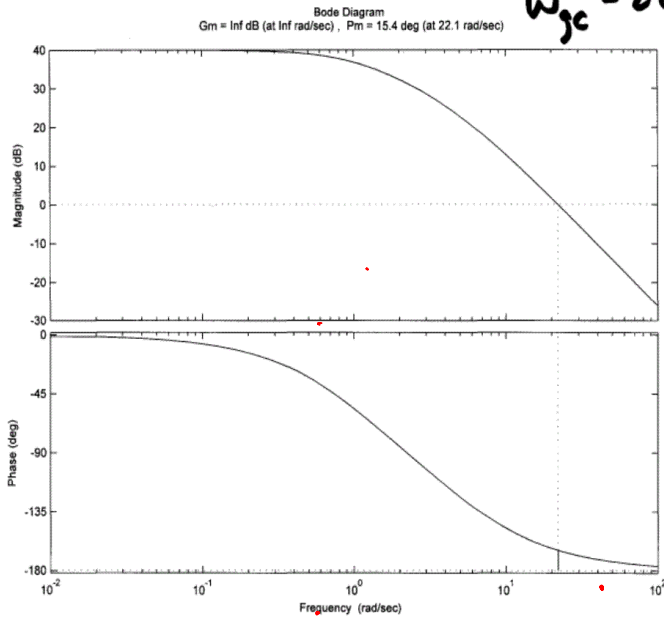
Note that we are not meeting $K_I > 0.1$ imposed by ess requirements.

Plots with $k=1$

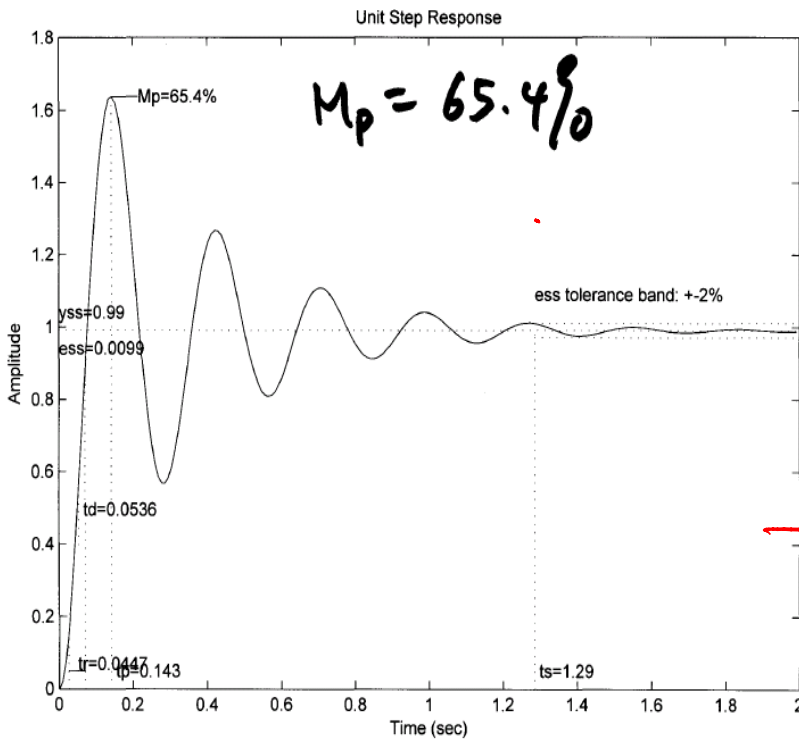
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$PM = 15^\circ$
 $\omega_{gc} = 22$



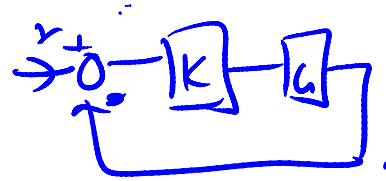
Bode plot



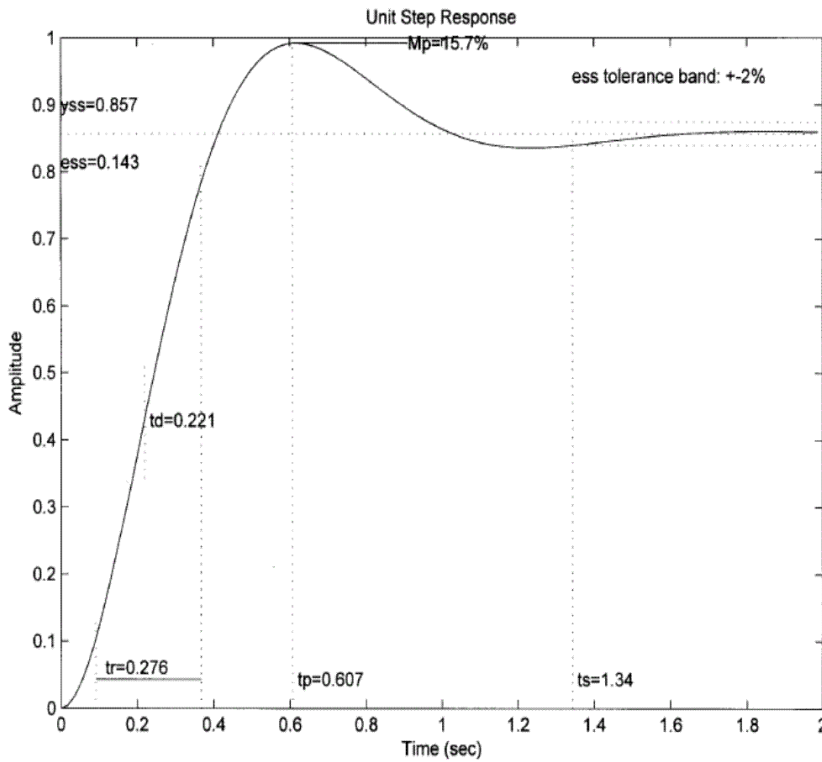
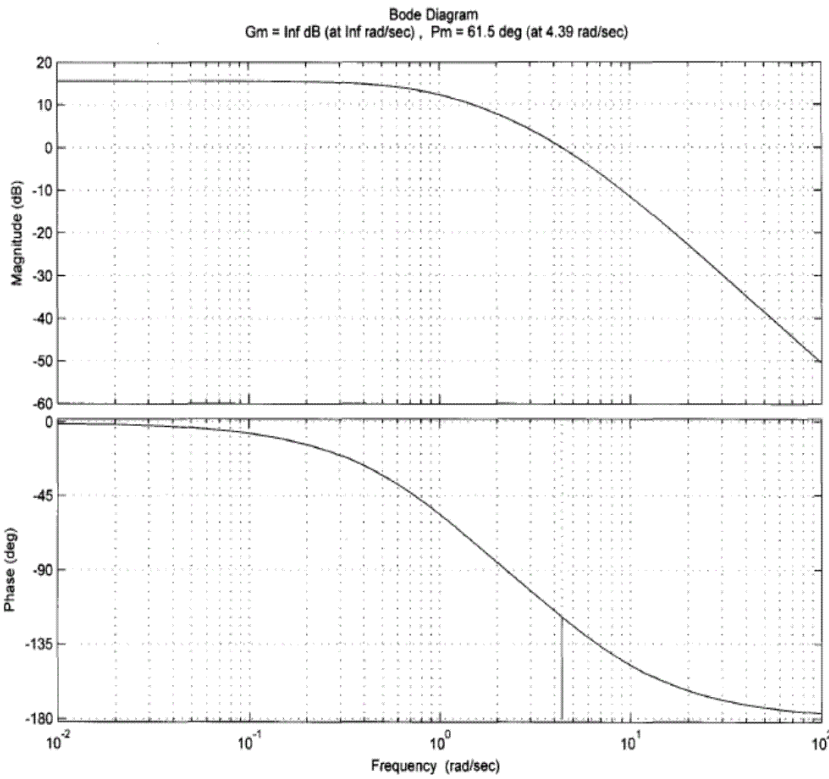
Step response

Closed-loop characteristics with designed PI

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$$K = 0.06 + 0.044 \frac{1}{s}$$



Note that
 $K_r = 4.4$
instead of
10.

Iteration

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Note that $K_v \approx 4.4$ instead of 10

and therefore the ess requirement is not met.

→ There is some leeway for further iterating on K_p and K_I .
Note that desired overshoot is 0.16 and with the controller

$$0.06 + \frac{0.066}{5}$$

the overshoot is 0.15

Thus, there is "not much" leeway.